

AMERICAN UNIVERSITY OF BEIRUT
FACULTY OF ENGINEERING AND ARCHITECTURE
EECE 460 Control Systems
Spring 2003-2004
Quiz II
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Name :

1.5 hours.

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Total of 100 points

Open Book Exam, 2 pages

YOU MUST RETURN THIS EXAM WITH YOUR ANSWER BOOKLET

Problem 1 (30 points):

The space station attitude control dynamics has the plant transfer function given by $G(s)$. Design a digital controller to have desired closed loop natural frequency around 0.3 rad/sec and damping ratio of 0.7 using emulation.

Assume that the supplied Continuous controller is $D(s)$ and the sampling frequency is 6 rad/sec.

$$G(s) = \frac{1}{s^2} \quad \text{and} \quad D(s) = \frac{0.81(s+0.2)}{(s+2)}$$

1. Is the sampling frequency fast enough for emulation? Justify.
 2. What are the desired transient specifications of the system?
 3. What are the corresponding s-plane desired poles?
 4. Design $D^*(z)$ using Zero Pole Matching discretization.
 5. Derive the equivalent algorithm (difference equation) for microprocessor coding.
 6. Is your derived routine implementable? Justify, if not suggest a practical solution.
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Problem 2 (35 points):

The discrete equivalent linear time invariant state model of a second order system is given by the following:

$$\begin{aligned} X(k+1) &= A X(k) + B U(k) \\ Y(k) &= C X(k) \end{aligned}$$

The matrices are:

$$A = \begin{bmatrix} 0 & 1 \\ -1 & 1 \end{bmatrix} \quad ; \quad B^T = [0 \ 1] \quad ; \quad C = [2 \ 0]$$

- a) What are the poles of the system in the z-plane?
- b) Is the given system fully observable? Justify.
- c) Design a predictor estimator to have desired poles in the Z-plane located at 0. (Supply needed gain vector).

Problem 3 (35 points):

The discrete equivalent (T is 0.1 sec) linear time invariant state model of a continuous second order system is given by the following:

$$\begin{aligned} X(k+1) &= A X(k) + B U(k) \\ Y(k) &= C X(k) \end{aligned}$$

The matrices are:

$$A = \begin{bmatrix} 1 & -1 \\ 0 & 1 \end{bmatrix} \quad ; \quad B^T = [1 \ 1] \quad ; \quad C = [1 \ 2]$$

The desired given reference state vector is given by $X_d(k)$ different from zero and corresponds to a given command input signal $U_c(k)$. Define the state error by $e(k) = X(k) - X_d(k)$.

- a) Is the given system fully controllable? Justify.
- b) Design a feedback gain matrix G such that $U(k) = U_c(k) - Ge(k)$ will force the closed loop system error to go to zero with desired poles in the Z-plane located at 0.4 and 0.6.
- c) What are the obtained transient step response specifications of the closed loop system in the continuous domain?